

iNAT-M200/TLN • iNAT-M200/xLN

Precise MEMS Based Inertial Navigation System with integrated INS/GNSS/xxx Data Fusion

iNAT-M200/xLN¹ is a member of the advanced iNAT series (iMAR Navigation and Timing) and one of the smallest powerful MEMS based INS/GNSS inertial navigation, measurement, surveying and control systems on the market for applications on the surface (land/sea) and in the air. It provides all kinematic measurements like acceleration, angular rate, attitude, true heading, velocity and position of the target vehicle in real-time with an data update rate of up to 250 Hz.

- robust, compact, light weight system, ~850 grams
- based on high grade MEMS Gyro, Accel technology and up to multi-frequency / multi-constellation GNSS with optional dual-antenna heading and RTK support; gyros highly resistant against vibration impacts
- integrated GNSS engine, up to RTK all frequ./const. (4 types of engines available: /TLN, /SLN, /RLN, /MLN)
- options for high/low range angular rate (-HRR/-LRR) and high range acceleration (-HRA) available
- odometer / wheel sensor aiding capability
- output of angular rate, acceleration, attitude, true heading, CoG, velocity and position in realtime with up to 250 Hz (adjustable)
- several processing modes: Standard mode with 1 m position accuracy and RTK mode with 0.02 m position accuracy
- interfaces: UART RS232 & RS422 / CAN / Ethernet / USB for realtime data output and RS232 for DGPS/RTK correction input; odometer / VMS
- up to 128 GByte internal memory ("black-box")
- several versions with surveying grade GNSS, economic grade GNSS, standard noise as well as low-noise inertial sensors are available
- easy to use, easy to configure; powerful GUI

Depending on the application, environmental environment and required realtime accuracy, the data fusion includes INS, GNSS, VMS or any other external sensor providing position and/or velocity, standard deviation and time stamp.

In urban canyons often the number of observable satellites is quite limited and therefore the iNAT-M200/xLN supports an all GNSS constellation data fusion. The 42+ state Kalman filter processing provides a significant better and more robust position and velocity result compared to standard solutions.

For land vehicles additionally an odometer aiding capability is available as an option, the scale factor of the wheel sensor is estimated automatically.

The **<u>iNAT-M200/TLN</u>** provides system performance and system reliability which is required in standard tasks of navigation, guidance and control, mapping, vehicle motion dynamics testing, trajectory surveying and platform control tasks for cars, trucks, naval vessels, civil and military aircrafts etc.

The **iNAT-M200/MLN** provides the same features, but containing a cheaper commercial grade L1 GPS+GLONASS engine engine with less robust GNSS solution in difficult environment.



The iNAT-M200 is delivered with the MS Windows (or LINUX or MacOS alternatively) based configuration software <u>iXCOM-CMD</u>. This software allows to configure the output interfaces, furthermore all output data can be displayed and stored online on the user's notebook, tablet or process computer. It also allows powerful playback capabilities and provides data



export in many formats (csv, xml, GoogleEarth, InertialExplorer, GrafNav). With <u>iREF-GNSS</u>, iMAR also provides a GNSS reference station to provide RTK corrections for centimeter level accuracy on demand.

A powerful postproc software <u>iPosCAL-SURV</u> for batch processing is available to allow post-mission processing including a multi station GNSS correction data solution and a direct visualisation of the results in Google EarthTM.

iNAT-M200/TLN (multi-freqency GPS/GALILEO/GLONASS/BEIDOU) technical modifications reserved w/o notice



¹ Meaning of "x": the iNAT-M200/xLN can be delivered with 4 classes of integrated GNSS engines. Standard device is



Technical Data iNAT-M200/TLN and iNAT-M200/xLN (typical, rms):

	Rate	Acceleration	Attit./Heading	Position	Velocity	Height
Range ¹ : Bias Stability (AV): Bias (filtered ²): Bias day-to-day ³ :	± 400 °/s < 0.5 °/h 3 °/h < 0.07 °/s	± 10 g (opt. 30 g) < 0.06 mg < 1.5 mg < 6 mg	unlimited	unlimited	515 m/s (without export control)	unlimited
Angles (Attitude, Hdg.):		0.03° / 0.1° RP/Y ((INS / GNSS) ⁵	(after 40 a C	NCC automa) 5	
			0.05° / 0.12° RP/Y 0.1 deg heading fo	after 10 s G or 2 m baseline	e in dual-anten	na setup (/SLN-DA) ²
Position (horizontal plan	ne) ⁴ :	_for iNAT-M200/TLN for iNAT-M200/MLN	:+/- 0.1 m CEP (IN <i>I</i> :_ 1 % of DT CEP (wi	NS/GNSS RTH +/- 0.03 m C +/- 0.4 m C +/- 1.8 m C 0.15 % of D +/- 1.8 m C ith VMS, durin	Creal-time) ⁵ EP (INS/GNSS EP (INS/GNSS EP (INS/GNSS Γ CEP (with VM EP (INS/GNSS g GNSS outag	S RTK post-proc) ⁵ S with SBAS) S) MS, during GNSS outages) ⁶ S) jes) ⁶
Velocity:					_0.02 m/s (IN	S / RTK GNSS)
Noise: Resolution:	0.15 °/sqrt(hr) < 0.000'1 °/s	60 µg/√Hz < 10 µg	0.03 ° 0.001 °	< 0.01 m < 0.001 m	< 0.01 m/s < 0.001 m/s	
Linearity error: Scale factor error: Non-orthogonality:	< 0.01 % < 0.1 % < 0.5 mrad	< 0.05 % < 0.1 % < 0.5 mrad	< 0.1 % < 0.1 %			
INS / GNSS / ODO proc. Internal GNSS Engine:	integrated adv version /TLN: version /SLN: version /RLN:	vanced 42+ state INS/C high performance all fr performance up to all f commercial multi frequ	GNSS/+ extended Kalr equencies / all conste frequency / constellatio uencies / multi constell	man filter data t llation RTK GN on geodedic cla lation RTK GN	fusion (GPS, G/ ISS engine (sing ass RTK GNSS SS engine (sing	ALILEO, GLONASS, BEIDOU) gle & dual antenna available) engine (single or dual antenna) le & dual antenna available)
Data Processing Rate: Data Output Rate: Synchronisation: Output (options):	version /MLN: economic grade L1 GPS+GLONASS, SBAS, Beidou, QZSS engine up to 250 Hz; PPS timing accuracy better 10 ns integer divisor of 250 Hz; all data available in real time, latency < 3 ms, jitter < 1 ms PPS_OUT (RS422 level, latency < 1 µs); 2x EVENT_IN (RS422 or TTL level, latency < 2 ms) USB, 2 x CAN, 4 x UART RS232/422, Ethernet 100 Mbit/s, NMEA183, ARINC825, TCP/IP, UDP, NTRIP cas- ter with PTCM104 ray 3 (can serve as a CNSS reference station): NTR Time Server (since HW ray, 5)					
Inputs:	DGPS/RTK c	prrection data from bas	e station, if available ((RS232);		<i>Je Hivi Tev. J</i>
Data Latency: Connectors: Integrated Data Storage: Graphical User Interface:	odometer / VMS (A or A/B at RS422 level) as an option < 11.3 ms (sampling accuracy better 1 µs, time-stamped according to PPS; jitter < 1 ms) MIL-C-38999 III (data), SMA (antenna), M12 (Ethernet) 32 GByte non-vlatime memory, optional up to 128 GByte (lasts for several days continuous data sampling as "black-box") MS Windows or LINUX or MacOS based software <u>iXCOM-CMD</u> for configuration, visualization, data recording data converting and playback operation.					
Power Supply:	934 V DC, t approx. 8.5	934 V DC, two independent and isolated inputs available; reverse an overvoltage protection; approx. 8.511 W (dep. on options); < 14 W for < 1 sec after power-on				
Temperature; MTBF:	-40+71 °C (-40+71 °C (outer case temperature) operating, -4085 °C storage; 49'000 hrs (AUC, Airborne Uninhabited Cargo, 25 °C)				
Shock, Vibration, Altitude g / g ² depenent gyro drift Mass, size; IP:	60 g, 11 ms, 102'000 Hz 5 g rms (endurance); 102'000 Hz 2 g rms (operational); 60'000 ft < 1 °/h/g / 0.06 °/h/g² [at 5 g / 1'000 Hz] (internally compensated) approx. 900 grams , approx. 102 x 138 x 65 mm; IP67 environmental protection					
Operational Support:	Airborne, Ground (with and without odometer), Sea, Subsea; ZUPT (auto or on demand), open interface to feed in application specific aiding information (position, velocity, attitude, heading, standard deviations, time stamp)					
Deliverables:	- MEMS base - iXCOM-CMI	- MEMS based INS with integrated GNSS receiver, GNSS antenna, cable set - iXCOM-CMD_MS Windows or LINUX or MacOS based GUI / HMI software				
PartNumber:	standard vers	ion: P/N 00190-000	01-xxxx (rev. 3) [for	advanced acc	uracy see the "in	mproved" version]
Options:	 L1 up to all-frequency / all constellation RTK accuracy of the integrated GNSS receiver dual-antenna GNSS based true heading (iNAT-M200/SLN-DA, iATTHEMO-C) allows heading determination even at standstill conditions -> typ. 0.2° at 1 m baseline odometer (VMS) interface for velocity aiding during longer GNSS outages (position error is then correlated to wheel sensor performance, typically 0.1 % longitudinal error of distance travelled) interface to iMAR's iDMN Dynamic Mesh Network for Swatm Communication & Control specific algorithms (also customized solutions) 					
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Option: iNAT-M200/SLN-HRR for high rate range tasks:	up to 1'200 deg/s
Option: iNAT-M200/SLN-HRA for high range acceleration tasks:	up to 30 g [then it might require an export license]
Option: iNAT-M200/SLN-HRS for high range speed tasks:	> 515 m/s (requires an export license)
² after algorithm converging under motion with GPS aiding at const. temp	perature
³ values without sufficient INS/GNSS data fusion conditions; the bias are	estimated / compensated during GNSS aiding under motion
automatically (Kalman filter); iNAT-M200-SLN provides 10 deg/hr bias	stability for several hours duration at const. temperature
GNSS based altitude deviation is about 1.5 times of GNSS based horiz	contal error; position error of iNAT-M200/MLN: < 2 m (CEP)
⁵ dependent on trajectory and satellite constellation: in challenging enviro	onments (e.g. urban canvons) use iNAT-M200/TLN

^o dependent on trajectory and satellite constellation; in challenging environments (e.g. urban canyons) use INAT-M200/TLN Hint: Under best GNSS and motion conditions the accuracy is 2 cm CEP.
 ⁶ Position error in relation to distance travelled (DT) during GNSS outages (requires a vehicle motion sensor / wheel sensor), after suffic. GNSS

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