



iNAT-M300/TLE-SP1

Precise MEMS Based Inertial Navigation System with integrated advanced INS/GNSS/xxx Sensor Data Fusion

The **iNAT-M300/xLE¹** is a member of the advanced iNAT series (iMAR Navigation and Timing) and one of the smallest powerful MEMS based INS/GNSS inertial navigation, measurement, surveying and control systems on the market for applications on the ground / land / rail, at sea and in the air. It provides all kinematic measurements like acceleration, angular rate, attitude, true heading, velocity and position of the target vehicle in real-time with an data update rate of up to 500 Hz.

- robust, compact, light weight system, ~850 grams
- based on high grade MEMS Gyro, Accel technology and up to multi-frequency / multi-constellation GNSS with optional dual-antenna heading and RTK support; gyros highly resistant against vibration impacts
- integrated GNSS engine, up to RTK all frequ./const. (3 types of engines available: /TLE, /SLE, /RLE)
- iNAT-M300/TLE-SP1: standard performance version
- · odometer / wheel sensor aiding capability
- output of angular rate, acceleration, attitude, true heading, CoG, velocity and position in realtime with up to 500 Hz (adjustable)
- several processing modes: Standard mode with 1 m position accuracy and RTK mode with 0.02 m position accuracy (rms)
- interfaces: UART RS232 & RS422 / CAN / Ethernet / USB for realtime data output and RS232 for DGPS/RTK correction input; odometer / VMS
- up to 128 GByte internal memory ("black-box")
- several versions with surveying grade GNSS, economic grade GNSS, standard noise as well as low-noise inertial sensors are available
- easy to use, easy to configure; powerful GUI with wizard; drivers for C++, Python and ROS 2 node

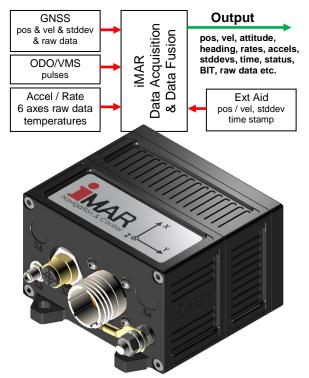
Depending on the application and ambient conditions, the integrated sensor data fusion includes INS, GNSS, VMS or any other external sensor providing position and/or velocity, standard deviation and time stamp.

In urban canyons often the number of observable satellites is quite limited and therefore the iNAT-M300/TLE supports an all GNSS constellation data fusion. The 42+ state Kalman filter processing provides a significant better and more robust position and velocity result compared to standard solutions.

For land vehicles additionally an odometer aiding capability is available as an option, the scale factor of the wheel sensor is estimated automatically.

The **<u>iNAT-M300/TLE</u>** provides system performance and system reliability which is required in general tasks of navigation, guidance and control, mapping, vehicle motion dynamics testing, trajectory surveying and platform control tasks for cars, trucks, naval vessels, civil and military aircrafts etc.

The **iNAT-M300/RLE** provides the same features, but containing an economic grade GNSS engine with less robust GNSS solution in difficult environment.



The iNAT-M300 is delivered with the MS Windows (or LINUX or MacOS alternatively) based configuration software <u>iXCOM-CMD</u>. This software allows to configure the output interfaces, furthermore all output data can be displayed and stored online on the user's notebook, tablet or process computer. It also allows powerful playback capabilities and provides data export in many formats (csv, xml, GoogleEarth, iPosCAL, InertialExplorer, GrafNav). With <u>iREF-GNSS</u>, iMAR also provides a GNSS reference station to provide RTK corrections for centimeter level accuracy on demand.

A powerful postproc software <u>iPosCAL-SURV</u> for batch processing is available to allow post-mission processing including a multi station GNSS correction data solution and a direct visualisation of the results in Google EarthTM.

iNAT-M300 is a leading solution of its class also for applications in GNSS denied environment.



GPS/GALILEO/GLONASS/BEIDOU)

iNAT-M300/TLE (multi-freqency multi-constellation

technical modifications reserved w/o notice

¹ Meaning of "x": the iNAT-M300/**x**LE can be delivered with 3 classes of integrated GNSS engines. Standard device is





Technical Data iNAT-M300/TLE-SP1 and iNAT-M300/xLE-SP1 (rms):

	Rate	Acceleration	Attit./Heading	Position	Velocity	Height
Range ¹ : Bias Stability (AV): Bias (filtered²):	± 450 °/s < 1.2 °/h 6 °/h	± 8 /16 g < 0.02 mg < 1 mg	unlimited	unlimited	515 m/s (without export	unlimited
	< 0.1 °/s	< 3 mg			control)	
Angles (Attitude, Hdg.):		-	0.05° / 0.1° RP/Y	(INS / GNSS)		
			0.06° / 0.12° RP/Y	(after 10 s GN		
	4.		0.1 deg heading for			
Position (horizontal plane	ə) *:	tor INA I-M300/ I LE	-SP1:	+/- 0.03 m C +/- 0.45 m C +/- 1.8 m C	EP (INS/GNSS EP (INS/GNSS EP (INS/GNSS	RTK post-proc) with SBAS)
Velocity:					0.02 m/s (IN	S / RTK GNSS)
Noise / ARW: Resolution:	0.08 °/√hr ⁶ < 0.000'1 °/s	60 µg/√Hz < 10 µg	0.03 ° 0.001 °	< 0.01 m < 0.001 m	< 0.01 m/s < 0.001 m/s	
Linearity error: Scale factor error: Non-orthogonality:	< 0.05 % < 0.2 % < 0.2 mrad	< 0.1 % < 0.1 % < 0.2 mrad	< 0.1 % < 0.1 %			
NS / GNSS / ODO proc. Internal GNSS Engine:	version /TLE version /SLI	E-x: high performance a E-x: performance up to	all frequencies / all cons all frequency / constella	stellation RTK (ation geodedic	GNSS engine (s class RTK GNS	ALILEO, GLONASS, BEIDOU) single & dual antenna available) SS engine (single or dual antenna) ingle & dual antenna available)
Data Processing Rate: Data Output Rate: Synchronisation: Dutput (options):	1500 Hz; PPS_OUT (USB, 2 x C/	all data available in rea RS422 level, latency < AN, 4 x UART RS232/4		itter < 1 ms RS422 or TTL l s, NMEA183, A	evel, latency < 2 RINC825, TCP	2 ms) /IP, UDP, NTRIP caster with
nputs: Data Latency:	DGPS/RTK	correction data from ba		(RS232); odom	neter / VMS (A o	Real-Time-Clock (RTC) as option or A/B at RS422 level) as an optio
Connectors:			nna), M12 (Ethernet); o			67)
Integrated Data Storage: Graphical User Interface: Power Supply:	32 GByte no MS Window 1034 V D0	on-vlatime memory, opt is or LINUX or MacOS C, two independent and	ional up to 128 GByte (based software <u>iXCOM</u> I isolated inputs availab	(lasts for severa - <u>CMD</u> for confi ole; reverse pol	al days continue guration, visual	ous data sampling as "black-box") ization, data recording, playback
Temperature; MTBF: Shock, Vibration, Altitude Qualification: g / g ² depenent gyro drift: Mass, size; IP:	designed to meet MIL-STD-810G, MIL-STD-461G, NIL-STD-704F and partially DO160G					
Operational Support:	Airborne, Ground (with and without odometer), Sea, Subsea; ZUPT (auto or on demand), open interface to feed in application specific aiding information (position, velocity, attitude, heading, standard deviations, time stamp)					
Deliverables:	 MEMS based INS with integrated GNSS receiver, GNSS antenna, cable set ROS-2 node, Python driver, SDK for C++ available for easy integration iXCOM-CMD MS Windows or LINUX or MacOS based GUI / HMI software 					
PartNumber:	standard per		-M300/TLE-SP1-DA -M300/RLE-SP1-DA -M300/SLE-SP1-DA	P/N 0019	93-000E2-0517 (93-000E2-0417 (93-000E2-0317 (,
	low noise ve	iNAT iNAT	-M300/TLE-LN1-DA -M300/RLE-LN1-DA -M300/SLE-LN1-DA NSS antenna setup: P/	P/N 0019 P/N 0019		dual-antenna) dual-antenna, incl. L-Band)
Features and Options:	 low noise dual-anten allows hea odometer correlated interface to 	version available: iNA na GNSS based true h ding determination eve (VMS) interface for velo to wheel sensor perforn b iMAR's <u>iDMN</u> Dynami	T-M300/TLE-LN1 (range eading solution availab n at standstill condition ocity aiding during longe mance, typically 0.1 % l c Mesh Network for Sw UTPUT for marine / nav	ge: +/- 450 deg, le (i NAT-M300 s -> typ. 0.2° a er GNSS outag longitudinal err vațm Communi	/s, ARW 0.06 di /xLE-SP1-DA) t 1 m baseline es (position err or of distance tr cation & Contro	eg/sqrt(hr)) or is then ravelled) ol

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¹ Option: **iNAT-M300/xLE-HRS** for high range speed tasks: > 515 m// ² after algorithm converging under motion with GNSS aiding at const. temperature > 515 m/s (requires an export license)

³ values without sufficient INS/GNSS data fusion conditions; the bias are estimated / compensated during GNSS aiding under motion

automatically (Kalman filter); iNAT-M300-xLE provides 3 deg/hr bias stability over longer duration at const. temperature ⁴ GNSS based altitude deviation is abut 1.5 times of GNSS based horizontal error

⁵ Position error in relation to distance travelled (DT) during short GNSS outages (requires a vehicle motion sensor / wheel sensor), after suffic. GNSS